

# SLS-RFM\_23-02 MFSK for very low data rates

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- Background
- MFSK basics
- ESA study
- Way forward

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- ESA study
- Way forward

Deep-space missions under certain conditions (high Doppler rate, high phase noise or very low SNR) challenging to demodulate using coherent schemes.

-> non-coherent **MFSK** (Multiple Frequency-Shift Keying) can be a solution for communication in these conditions.

**NASA/JPL** has already used MFSK:

- Mars rovers: EDL (DTE X-band)
- Juno: Deep-Space Manoeuvres and Jupiter Orbit Insertion
- Europa Clipper, ...?
- *There is also a simpler “beacon tone” option to transmit basic spacecraft status*

**ESA** has not yet used MFSK, but:

- Now implemented in next gen deep-space transponders
- Ongoing study:
  - analyse use cases and solutions
  - implement MFSK receiver
- *DOR tones can be used as beacon tones*

**Other agencies?**

The topic has been for some time in the **RFM charter**:

*18) Study modulation technique and position for pilot symbols of high order modulations used in conjunction with the codes of CCSDS 131.0-B-2*

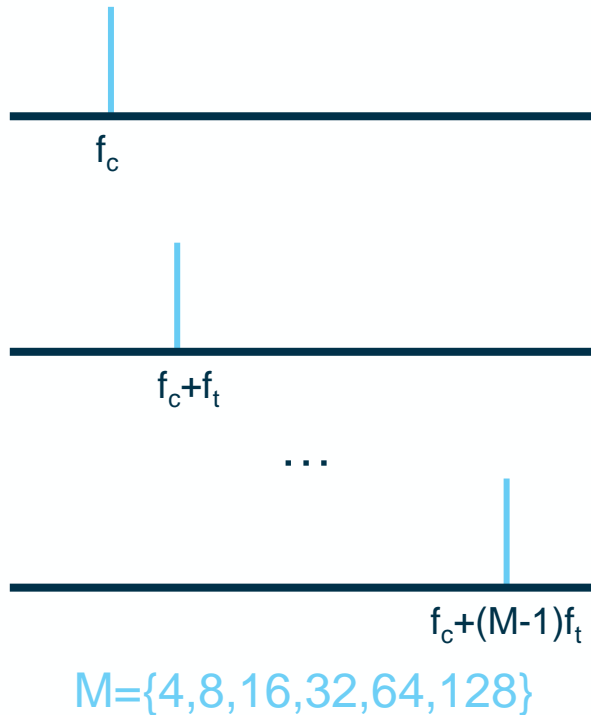
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- **Entry, Descent and Landing** (high Doppler rate uncertainty)
- **Solar conjunction** (high phase and amplitude scintillation)
- **Safe/Survival mode** (very low SNR)

# MFSK modulation

**Classical MFSK:** information resides in the tone frequency

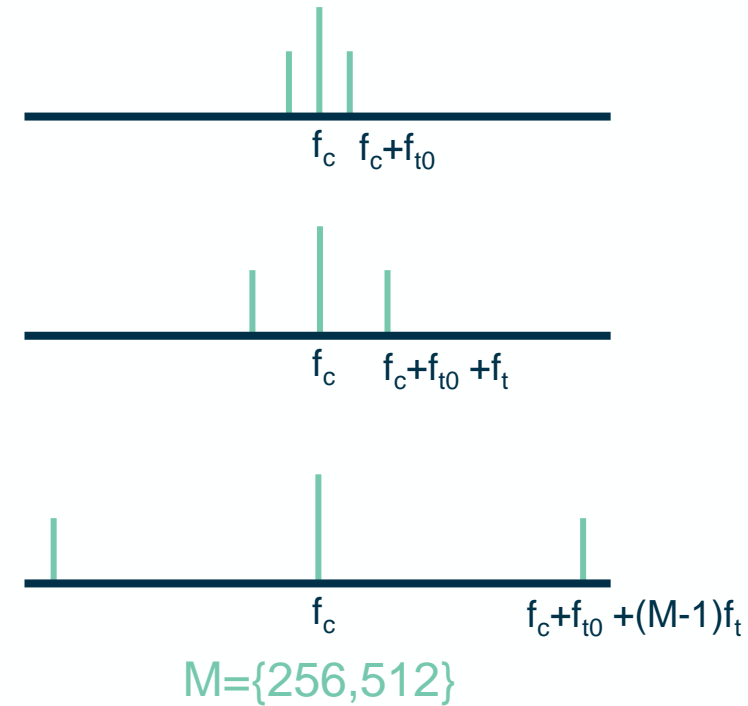
$$X_i^{\text{MFSK}}(t) = \sqrt{2P_T} \cos(2\pi(F_c^0 + F_i)t + \theta)$$



**Special MFSK:** information resides in the frequency separation between carrier and tone

$$X_i^{\text{SMFSK}}(t) = \sqrt{2P_T} \cos(2\pi F_c^0 t + \Delta \text{sqr}(2\pi F_i t) + \theta)$$

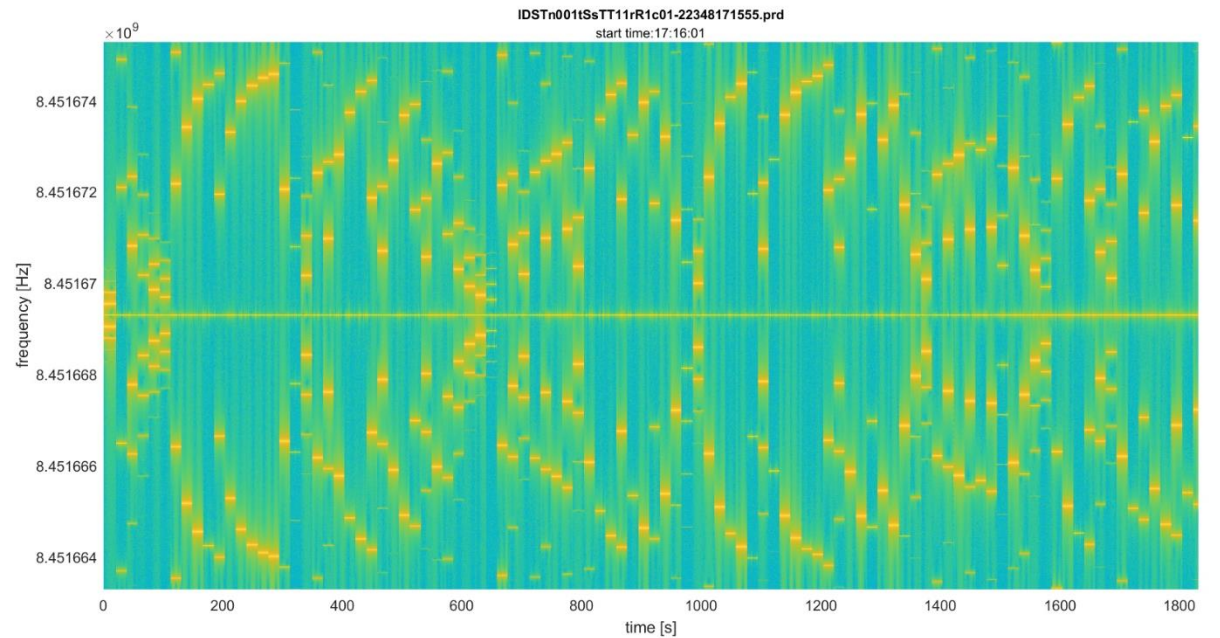
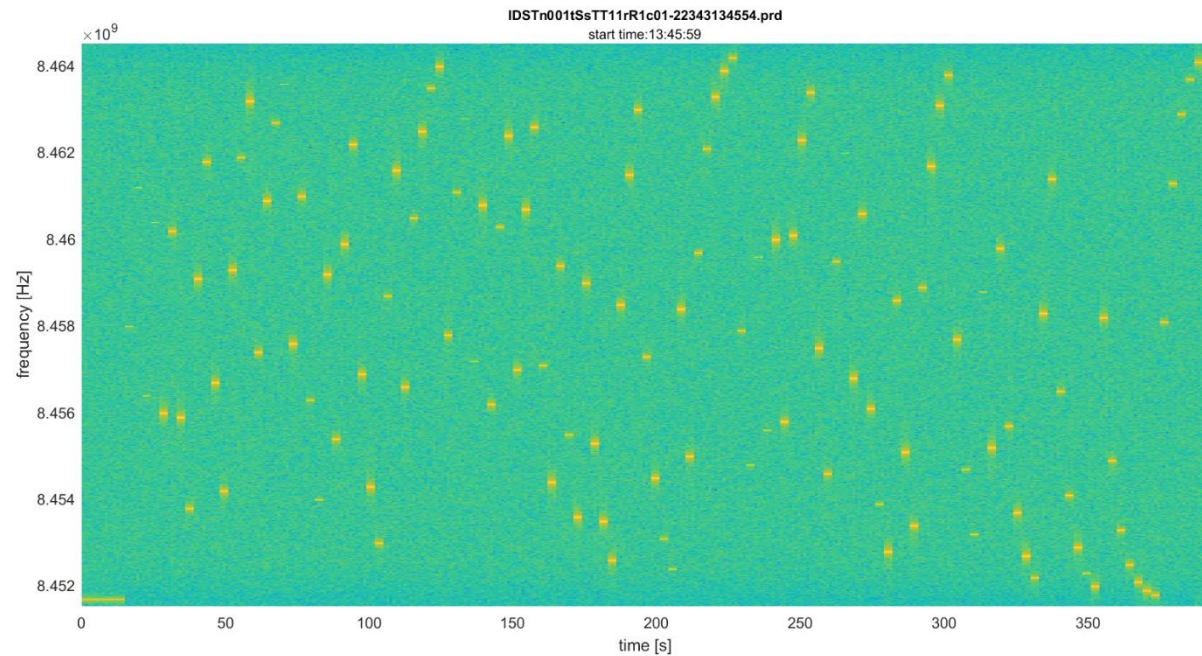
(sinewave also possible)



used by JPL

**Classical MFSK:** information resides in the tone frequency

**Special MFSK:** information resides in the frequency separation between carrier and tone



[MFSK signal from IDST transponder, OL recordings performed at ESOC, Dec 2022]

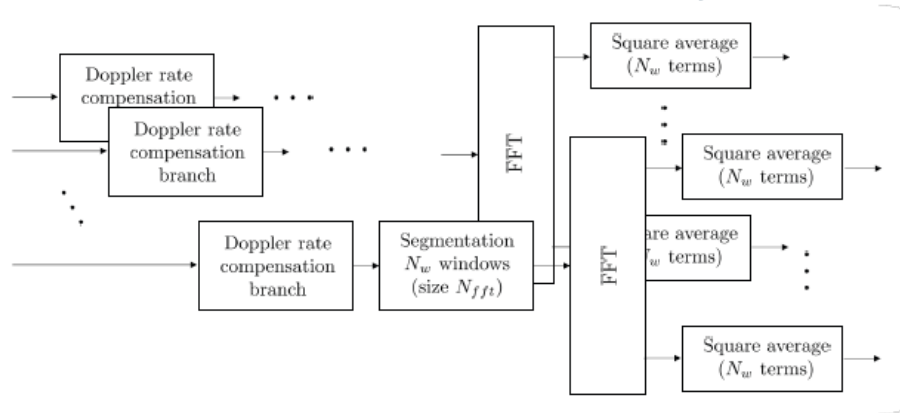


Modulation parameters:

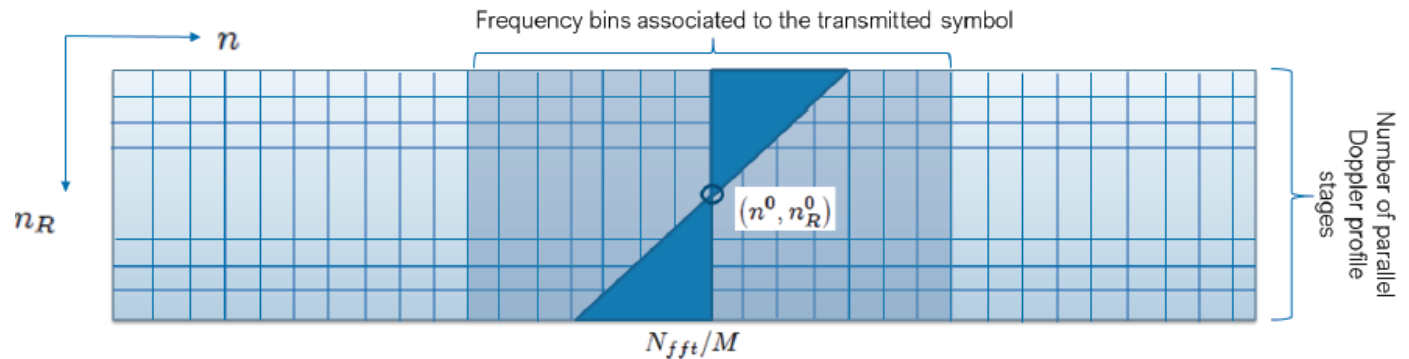
- Modulation type [ classical MFSK / special MFSK ]
  - Modulation order:  $M$
  - Tone duration:  $T_S$
  - Separation between tones:  $f_T$
  - Separation carrier to 1<sup>st</sup> tone:  $f_{T0}$
  - Tone waveform [ sine / square ]
  - Modulation index:  $\Delta$
- } only applicable to special MFSK

In classical MFSK, separation between tones ( $f_T$ ) must be sufficient to cover the maximum Doppler error  
In special MFSK, only the separation of the first tone ( $f_{T0}$ ) must be higher than the maximum Doppler error

- Signal recorded in open-loop, ideally Doppler-precompensated
  - Sampling frequency must cover all possible tones + Doppler uncertainty
  - Possibility to array multiple antennas to improve SNR -> potential cross-support
- Demodulation with quasi-real-time tool or offline tool
  - Depending on operational needs and processing power
  - Post-processing with finer Doppler correction might further improve detection
- Tone detection based on 2-D FFT
  - search for best **frequency** match and best **frequency rate** match



[from ESA MFSK study] Receiver structure for classical MFSK



[from ESA MFSK study] Detection of a symbol in the frequency / Doppler rate grid

So far, JPL use of MFSK seems limited to indication of spacecraft status or events.

## Can we use it to transmit TM according to CCSDS?

Coding:

- Codes in 131.0-B are designed for AWGN and coherent demodulation, we cannot expect same coding gain
- Lack of references of performance for 131.0-B codes with MFSK
- Frame duration for standard lengths become very long
  - Risk of losing a full frame might not be acceptable, in particular for **EDL** (*7 minutes of terror!*)

Data link and upper layers:

- Overhead increases transmission time
- But without frame structure, no standard way for cross-support (SLE)

- To our knowledge, no spacecraft has used MFSK for uplink so far
  - But there seems to be interest for future missions
- Applicable in the same scenarios as for downlink
- Simple to implement on ground station, but complexity is transferred to spacecraft
- Coding and upper layers?

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- ESA-funded study, carried out by research institutes: CTTC + Ceit
- Inputs from previous ESA studies on EDL, solar conjunction
- Focused on downlink scenarios
  
- Main objectives:
  - Analysis of scenarios and selection of modulation and coding schemes
  - Implementation of MFSK receiver

- Worst-case scenarios considered during study:

	EDL (X band)	Solar conjunction (X-band)	Solar conjunction (Ka band)	Safe/Survival mode (X band)
Doppler frequency error	<±50 kHz	<±100 Hz	<±400 Hz	<±100 Hz
Doppler frequency rate error	<±700 Hz/s	<±1.5 Hz/s	<6 Hz/s	<1.5 Hz/s
S/N0	> 14.2 dBHz	≥ 30 dBHz	≥ 30 dBHz	≥ 6 dBHz
Amplitude scintillation index		< 0.85 <small>(Sun-Earth-Probe angle = 1 deg)</small>	< 0.4 <small>(Sun-Earth-Probe angle = 0.5 deg)</small>	

# ESA MFSK study - preliminary results



Optimisation done for the worst-case scenarios to achieve max symbol rate with acceptable computational complexity, while keeping  $P_{fd} < 10^{-4}$

...this means correctly detecting 99.99% of the tones.

Maybe too harsh!

For M2020 EDL, requirement was 90%

[L. Mauger *et al.*, "Direct to Earth Communications Using MFSK Tones during M2020 Entry, Descent, and Landing," ]

Scenario	waveform	M	Tone separation	Tone duration	Modulation index	Symbol rate (approx.)	Complexity
Safe/survival mode	Classical MFSK	128	1 kHz	14 s	NA	0.5 sps	3.4e8 op/s
	Special MFSK sinewave	512	10 Hz	17 s	80 deg	0.5 sps	1.1e8 op/s
EDL	Classical MFSK	64	100 kHz	2.8 s	NA	2 sps	1.0e11 op/s
	Special MFSK sinewave	512	200 Hz	2.9 s	70 deg	3 sps	5.1e10 op/s
Solar conjunction X-band	Classical MFSK	128	50 kHz	0.11 s	NA	63 sps	7.9e7 op/s
	Special MFSK squarewave	512	10 kHz	0.16 s	70 deg	56 sps	1.3e8 op/s
Solar conjunction Ka-band	Classical MFSK	128	10 kHz	0.04 s	NA	160 sps	1.4e7 op/s
	Special MFSK squarewave	512	1 kHz	0.06 s	70 deg	150 sps	1.2e7 op/s

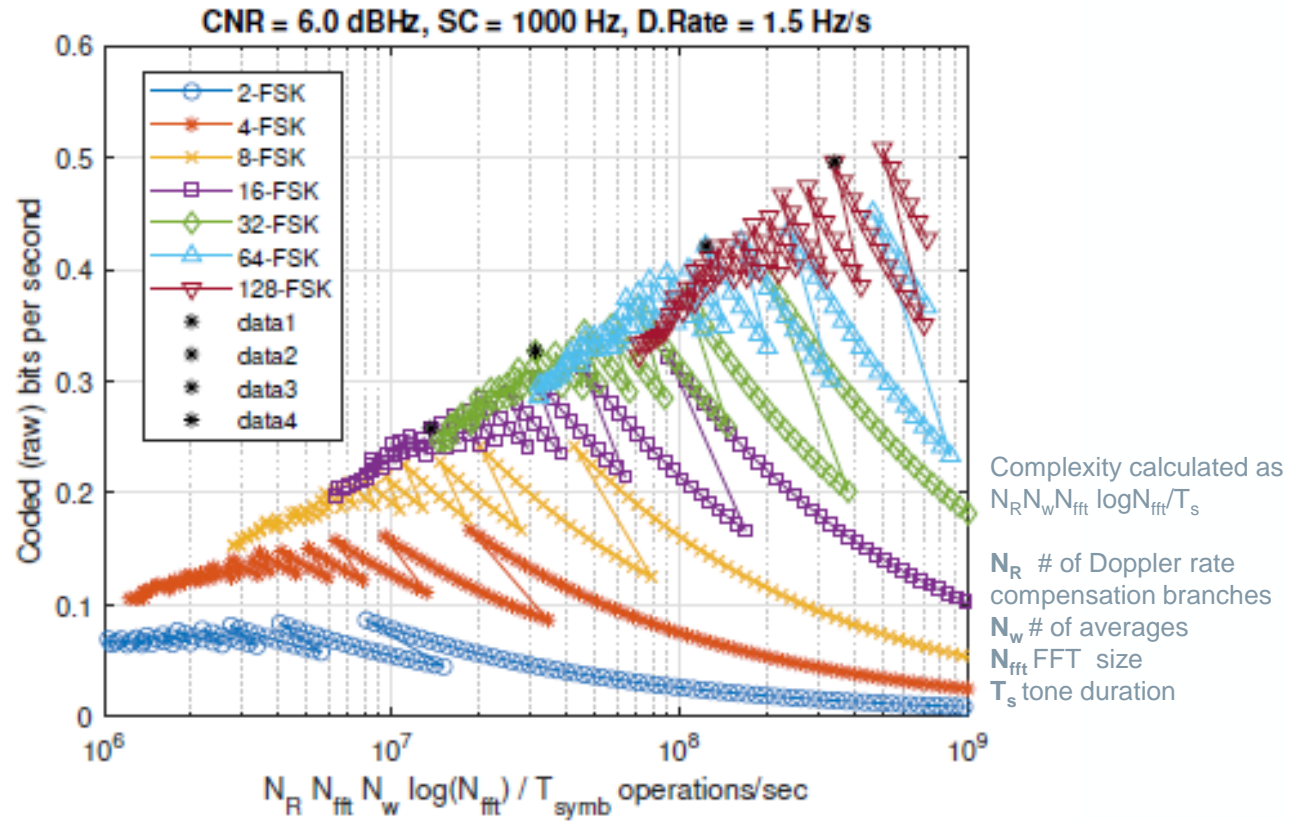
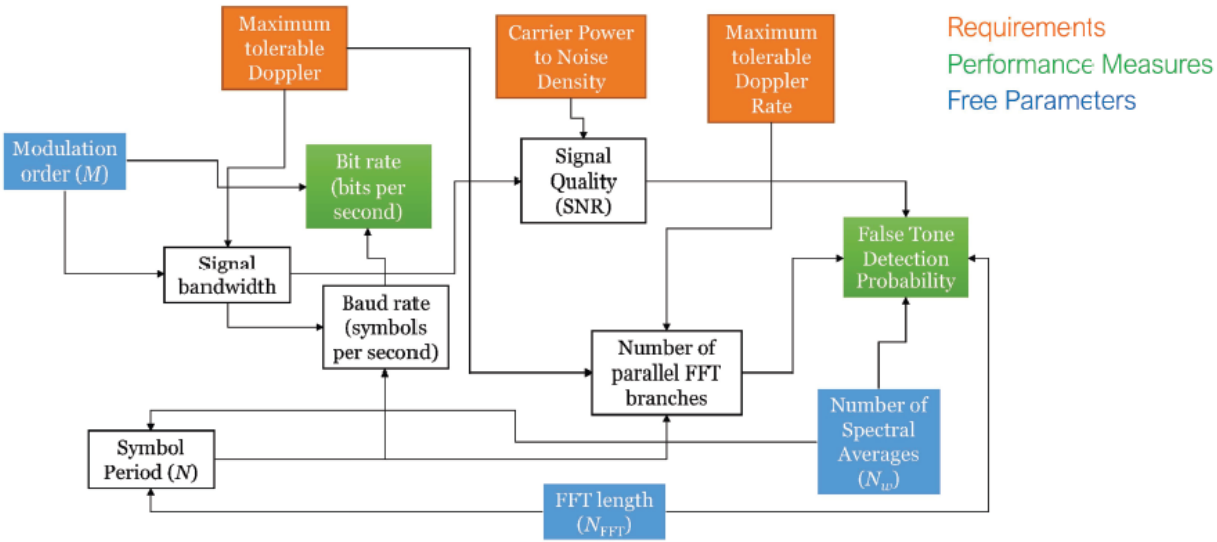
Relaxation of the Pfd to  $10^{-3}$  would allow increase to ~0.7sps (↑40% rate) with less complexity

Previous ESA study on EDL gave advantage to classical MFSK, but these results show opposite





Example of optimisation for safe/survival scenario



(b) Coded (raw) bit rate versus complexity.

[from ESA MFSK study]

For coded signals, soft demapper instead of hard output of FFT detector

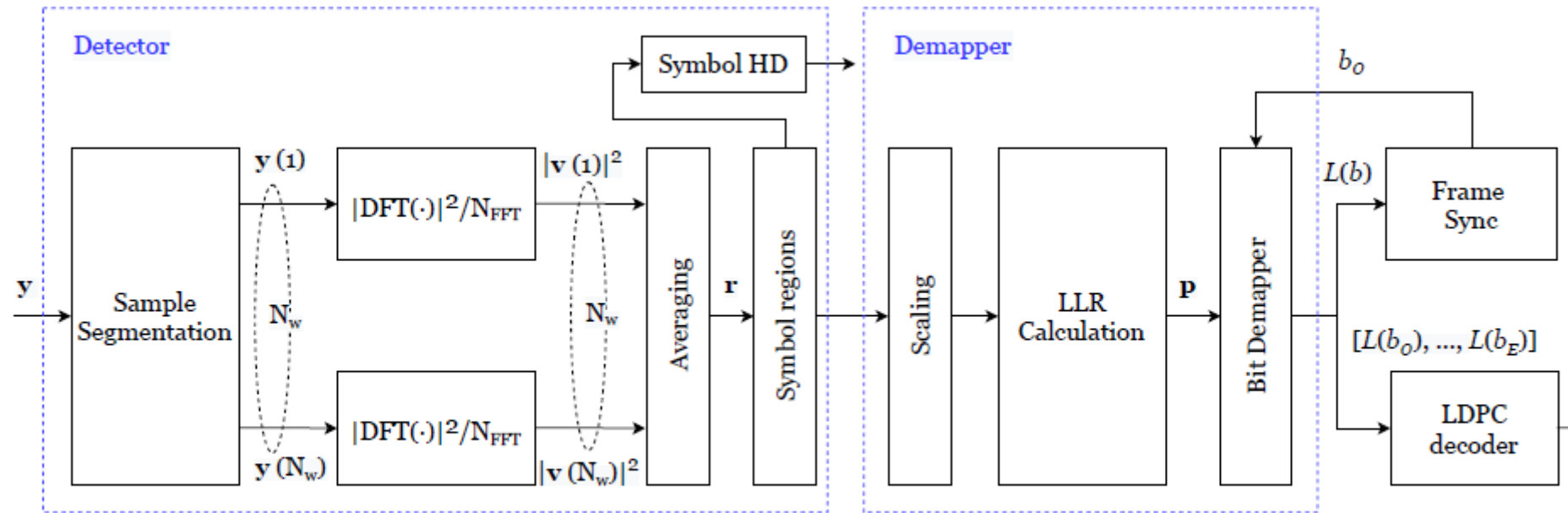


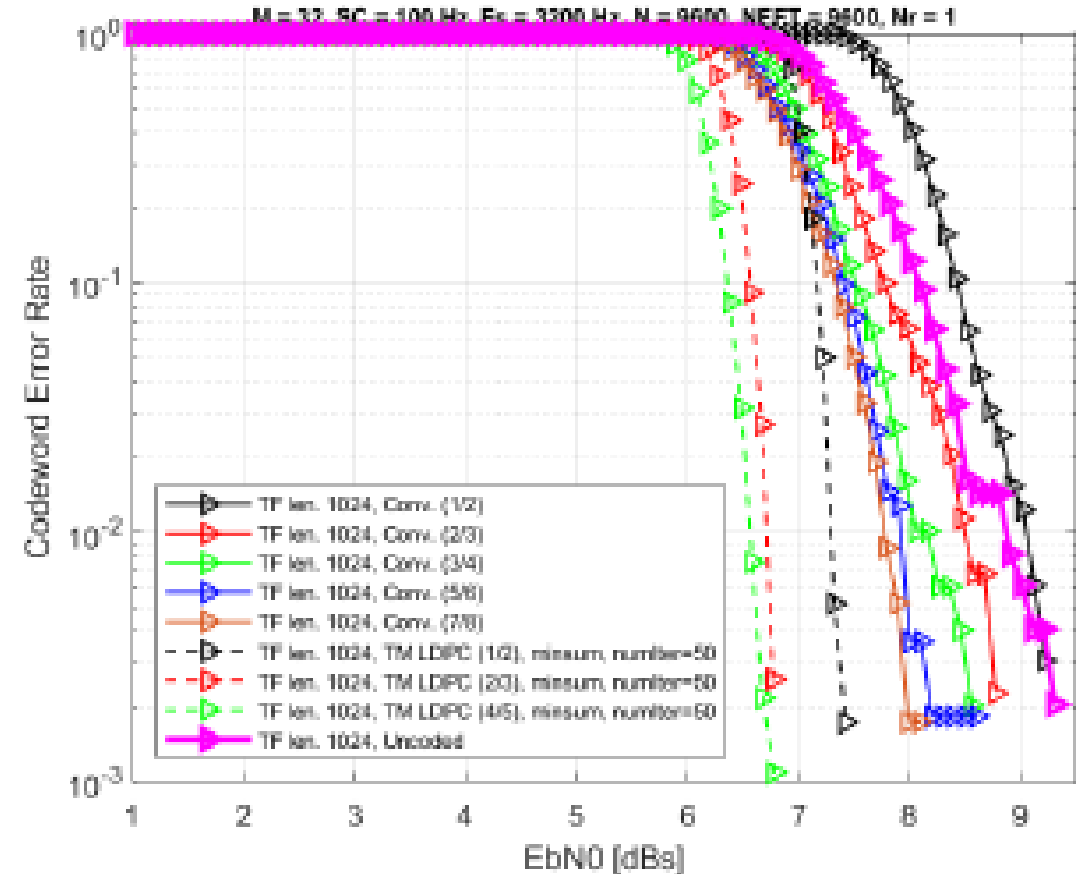
Fig. 3. Receiver architecture.

[J. Gómez-Vilardebó, X. Mestre, M. Navarro, J. F. Sevillano, R. Abelló and J. Quintanilla, "Non-Coherent Receiver Design for MFSK Modulations in Deep Space Missions," ]

# ESA MFSK study - preliminary results

- Analysis focused on LDPC and convolutional, otherwise frames too long
  - Shortest LDPC frame with  $k=1024$ ,  $r=4/5$  → 1344 symbols
    - 45 min at 0.5 sps (safe mode)
    - 7.5 min at 3 sps (EDL)
    - 21 s at 63 sps (solar conjunction X-band)
  - Shorter frames possible with convolutional, but worse performance
- End-to-end simulations including coding are intensive for current Matlab-based simulator, preliminary results based on **reduced scenario**:

	Optimised scenario for safe mode	Reduced scenario
<b>M</b>	128	<b>32</b>
<b>T<sub>S</sub></b>	~14 s	<b>3 s</b>
<b>N<sub>FFT</sub></b>	~600000	<b>9600</b>
<b>N<sub>w</sub></b>	3	<b>1</b>
<b>N<sub>R</sub></b>	201	<b>1</b>



- Far from results for AWGN, coherent (130.1-G)
- LDPC better than convolutional (for same length)
- Higher code rates are better!
  - LDPC **4/5** would be selected for this case
- More representative results expected at end of study

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- Wait for final results of ESA study... Fall 2023 or Spring 2024
- RFM WG position on adding MFSK to 401.0-B?
  - Only downlink or also uplink?
  - Should “beacon tones” also be included? (simpler version of MFSK)
- Coding and upper layers?
  - If no TM frames, should CCSDS define a new format for “very low rate information”?
  - Or investigate other codes outside 131.0-B?

- Satorius E, Estabrook P, Wilson J, Fort D. Direct-to-Earth Communications and Signal Processing for Mars Exploration Rover Entry, Descent, and Landing. *IPN Progress Report 42-153*. May 2003
- Soriano M, Finley S, Fort D, et al. Direct-to-Earth Communications with Mars Science Laboratory during Entry, Descent, and Landing. *IEEE Aerospace Conference*, 2013
- Dutta S, Soriano M, Optimizing Multiple Frequency-Shift Keying during Spacecraft Critical Events for Future Missions. *IEEE Aerospace Conference*, 2019
- D. Buccino *et al.*, "Detecting Juno's 'Heartbeat': Communications Support during Critical Events of the Juno Mission," *2020 IEEE Aerospace Conference*, Big Sky, MT, USA, 2020
- L. Mauger *et al.*, "Direct to Earth Communications Using MFSK Tones during M2020 Entry, Descent, and Landing," *2022 IEEE Aerospace Conference (AERO)*, Big Sky, MT, USA, 2022,
- Jet Propulsion Laboratory, Deep Space Network Services Catalog (820-100, Rev. H), June 6, 2022
- J. Gómez-Vilardebó, X. Mestre, M. Navarro, J. F. Sevillano, R. Abelló and J. Quintanilla, "Non-Coherent Receiver Design for MFSK Modulations in Deep Space Missions," *2022 9th International Workshop on Tracking, Telemetry and Command Systems for Space Applications (TTC)*, Noordwijk, Netherlands, 2022
- X. Mestre, J. Gómez, M. Navarro, J. F. Sevillano, R. Abelló and J. Quintanilla, "Non-coherent Receivers for Low Data Rate Transmissions under Weak Solar Scintillation," *2022 9th International Workshop on Tracking, Telemetry and Command Systems for Space Applications (TTC)*, Noordwijk, Netherlands, 2022
- Ongoing ESA-funded study, contracted to CTTC and Ceit (Spain) "Multiple Frequency-Shift Keying Modem for Very Low Data Rates"